Exercise-1

ONLY ONE OPTION CORRECT TYPE

SECTION (A): UNIVERSAL LAW OF GRAVITATION

SECT	IOIA (A) . GIAIVEINS	AL LAW OF GRAVII	AHON		
1.		tween ground and the ob exerted by earth on the o e of reference.			
2.	The weight of a body a (1) Zero (3) Same as on the sur	t the centre of the earth i face of earth	s - (2) Infinite (4) None of the	above	
3.	If the distance between (1) Is doubled (3) Is reduced to half	n two masses is doubled,	the gravitational (2) Becomes for (4) Is reduced to		
4.	The gravitational force vacuum is - (1) Zero (3) 6.675 × 10 ₋₁₁ newto		mass 1 kg each (2) 6.675 × 10 ₋₅ (4) 6.675 × 10 ₋₈		1 metre in
5.	Two particles of equal attraction. The speed of $\frac{1}{2R} \sqrt{\frac{1}{Gm}}$	of each particle is -		the action of their mutual g $(4) \text{ V} = \sqrt{\frac{4Gm}{R}}$	ravitationa
6.	Reason of weightlessn (1) Zero gravity (3) Zero reaction force		(2) Centre of ma (4) None	ass	
7.	The gravitational force (1) Sum of the masses (3) Gravitational consta		(2) Product of the		
8.		avitational force between	the parts?	parated by a certain distance.	. What ratio
	(1) $\frac{1}{3}$	$\frac{3}{4}$	(3) $\frac{1}{2}$	(4) $\frac{3}{3}$	
9.	On a planet (whose size	e is the same as that of	earth and mass 4	times to the earth) the energy planet is (g = 10m/sec ₂ on	
	(1) 16 J	(2) 32 J	(3) 160 J	(4) 320 J	
10.	The dimensions of univ (1) $[M_{-1}L_3T_{-2}]$	versal gravitational consta (2) [ML ₂ T ₋₁]	ant are : (3) [M ₋₂ L ₃ T ₋₂]	(4) [M ₋₂ L ₂ T ₋₁]	
SECT	ION (B) : GRAVITA	TIONAL FIELD AND	POTENTIAL		
4	If the change in the val	luc of 'a' at a baight h ab	ove the curfees o	of the earth is the same as a	ot a donth s

1. If the change in the value of 'g' at a height h above the surface of the earth is the same as at a depth x below it, then (both x and h being much smaller than the radius of the earth) -

(1)
$$x = h$$
 (2) $x = 2h$ (3) $x = \frac{n}{2}$ (4) $x = h_2$

2.			its mass is 1/80 time tha e earth, that on the surfa (3) g/6	t of the earth. If g represents the ce of the moon is (4) g/8	
3.			ohere of radius R, its de on the surface of the eart (3) 4πRg/(3G)	ensity in terms of G (constant of h) is $ (4) \ 4\pi RG/(3g) $	
4.	acceleration of that ob of earth and R is the ra	ject due to gravity at tha adius of earth)	t point is: (g = accelerat	ing mass is distributed uniformly, ion due to gravity on the surface (4) none of these	
	(1) g	(2) 2 g	(3) g/2	,	
5.	Altitude at which accel km)	eration due to gravity de	creases by 0.1% approx	imately : (Radius of earth = 6400	
	(1) 3.2 km	(2) 6.4 km	(3) 2.4 km	(4) 1.6 km	
6.		oden ball of the same ra to reach the ground is - (2) Exactly equal		a height 'h' in vacuum. The time	
	, ,	. , , , , , , , , , , , , , , , , , , ,	(3) Roughly equal	(4) Zero	
7.	The correct answer to above question is based on - (1) Acceleration due to gravity in vacuum is same irrespective of size and mass of the body (2) Acceleration due to gravity in vacuum depends on the mass of the body (3) There is no acceleration due to gravity in vacuum (4) In vacuum there is resistance offered to the motion of the body and this resistance depends on the mass of the body				
8.	When a body is taken from the equator to the po (1) Remains constant (3) Decreases		oles, its appearent weight - (2) Increases (4) Increases at N-pole and decreases at S-pole		
9.	A body of mass m is ta (1) Its mass increases (3) Its weight increases		eep mine. Then - (2) Its mass decreases (4) Its weight decrease		
10.	As we go from the equ (1) Remains the same (3) Increases	ator to the poles, the val	lue of g (2) Decreases (4) Decreases upto a I	atitude of 45°	
11.	Force of gravity is leas (1) The equator (3) A point in between		(2) The poles (4) None of these		
12.	Spot the wrong statement: The acceleration due to gravity 'g' decreases if - (1) We go down from the surface of the earth towards its centre (2)' We go up from the surface of the earth (3) We go from the equator towards the poles on the surface of the earth (4) The rotational velocity of the earth is increased				
13.		ntement from the followin astronaut moving in a sa (2) No gravity		(4) Free fall	
14.	If the earth suddenly s to gravity will be - (1) g/2	hrinks (without changing	g mass) to half of its pres	sent radius, the acceleration due (4) 2g	

15.

16.

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	(1) Half	(2) One-fourth	(3) One third	(4) No change
17.	At what height from the surface of earth -	e ground will the value o	of 'g' be the same as tha	t in 10km deep mine below the
	(1) 20 km	(2) 10 km	(3) 15 km	(4) 5 km
18.	At what distance from to on the surface (R = Rad		ne value of acceleration	due to gravity g will be half that
	(1) 2R	(2) R	(3) 1.414 R	(4) 0.414 R
19.		eration due to gravity at vity on the surface of ear		ere R is radius of earth and g is
	$\frac{g}{(x^2+y^2)^2}$	()	$\frac{g}{(x^2+y^2)^2}$	
	$(1) \left(1 + \frac{h}{R}\right)^2$	$(2) g \left(1 - \frac{2h}{R}\right)$	$(3) \left(1 - \frac{h}{R}\right)^2$	$(4) g \left(1 - \frac{h}{R}\right)$
20.	If the density of the ear	th is doubled keeping its	radius constant then acc	celeration due to gravity will be
	$(g = 9.8 \text{ m/s}_2)$			9 ,
	(1) 19.6 m/s ₂	(2) 9.8 m/s ₂	(3) 4.9 m/s ₂	(4) 2.45 m/s ₂
21.	The acceleration due to (1) $g_p < g_e$	gravity at pole and equal (2) $g_p = g_e = g$		(4) $g_p > g_e$
			<u>8</u>	<u> </u>
22.	The depth at which the	effective value of accele $3R$	ration due to gravity is $^{\prime}$	4 is - <i>R</i>
	(1) R	$\frac{3R}{4}$	$\frac{R}{2}$	$\frac{R}{4}$
23.	Two bodies of mass 10 the intensity of gravitati		one meter apart. At wha	at distance from 100 kg body will
	$\frac{1}{m}$		$\frac{1}{m}$	$\frac{10}{m}$
	(1) 9 "	(2) $\frac{10}{10}$ m	$\frac{1}{11} m$	(4) 11 "
24.	Figure show a hemisp intensity at point P will	_	form mass density. The	e direction of gravitational field
	, ,	o Z Z Z	a	
	(1) a	(2) b	(3) c	(4) d
25.	Two bodies of mass 10 the line joining them is	0₂ kg and 10₃ kg are lying	1 1m apart. The gravitation	onal potential at the mid-point of
	(1) 0	(2) -1.47 Joule/kg	(3) 1.47 Joule/kg	(4) -147 x 10 ₋₉ joule/kg
26.		s a time period T ₁ when ce, where R is the radius		and T_2 when taken to a height R of T_2/T_1 is:
	(1) 1	(2) $\sqrt{2}$	(3) 4	(4) 2
27.	Near earth's time period	d of a satellite is 4 h. Find	d its time period at distar	nce 4R from the centre of earth:

The moon's radius is 1/4 that of the earth and its mass is 1/80 times that of the earth. If g represents the

Radius of earth is around 6000 km. The weight of body at height of 6000 km from earth surface becomes

acceleration due to gravity on the surface of the earth, that on the surface of the moon is - (1) g/4 (2) g/5 (3) g/6 (4) g/8

28.	(1) 32 h The radius of the orbit of (1) 4.2 T	$ \begin{pmatrix} \frac{1}{8^{-3}\sqrt{2}} \\ \text{h} $ of a planet is two times the (2) 2.8 T	(3) $8^{-3} \sqrt{2} h$ nat of the earth. The time (3) 5.6 T	(4) 16 h period of planet is : (4) 8.4 T
29.	In case of earth: (1) field is zero, both at (2) potential is zero, bot (3) potential is same, bot (4) potential is maximum	th at centre and infinity oth at centre and infinity I	but not zero	
30.	What would be the ang $(g = 10\text{m/s}_2 \text{ and radius}$ (1) 1.25 × 10 ₋₃ rad/sec (3) 1.25 × 10 ₋₄ rad/sec		nat bodies lying on equat (2) 1.25 × 10 ₋₂ rad/sec (4) 1.25 × 10 ₋₁ rad/sec	or may appear weightless ?
31.		he earth have to rotate of sent will be (Take the equ		
SECT	ION (C) : GRAVITA	TIONAL POTENTIAL	ENERGY AND SEI	_F ENERGY
1.		rgy of the body is (g = gr	avity field at the surface	
	(1) mgR	(2) $\frac{4}{4}$ mgR	$\frac{1}{3} \operatorname{mgR}$	(4) $\frac{3}{3}$ mgR
2.	The change in potential is (R = Radius of earth		mass m is raised to a he	eight n R from the earth surface
		•	$\frac{n^2}{n^2+1}$ (3) mgR $\frac{n^2}{n^2+1}$	<u></u>
	(1) mgR	(2) nmgR	(3) mgR $n^2 + 1$	(4) mgR $n+1$
3.	If mass of earth is M, ra earth surface to infinity		al constant is G, then wo	ork done to take 1 kg mass from
	\overline{GM}	GM	2GM	GM
	(1) $\sqrt{2R}$	(2) <i>R</i>	(3) √ R	$(4) \overline{2R}$
4.		th velocity 10 km/s. If rad	lius of earth is R, then ma	aximum height attained by it will
	be- (1) 2R	(2) 3R	(3) 4R	(4) 5R
5.	What is intensity of grav (1) Gm/r ₂	vitational field at the cent (2) g	re of a spherical shell - (3) Zero	(4) None of these
6.		ody of 1 kg mass on a pl	anet is 100 m/sec. Grav	itational Potential energy of the
	body at the Planet is - (1) – 5000 J	(2) – 1000 J	(3) – 2400 J	(4) 5000 J
7.	The kinetic energy need (1) mgR/2	ded to project a body of r (2) 2 mgR	mass m from the earth su (3) mgR	urface (radius R) to infinity is - (4) mgR/4

The escape velocity of a sphere of mass m from earth having mass M and radius R is given by -

8.

	2GM	(2) 2 $\sqrt{\frac{GM}{R}}$		$\sqrt{\frac{2GMm}{R}}$		$\sqrt{\frac{GM}{R}}$	
	(1) \sqrt{R}	(2) 2 \sqrt{R}	(3)	\sqrt{R}	(4	\sqrt{R}	
9.	If g is the acceleration velocity for the body to	escape out of ea				dius of the earth, the esc	ape
	(1) gr	(2) $\sqrt{2gr}$	(3)	g/r	(4	l) r/g	
10.	For the moon to cease				-	to increase by a factor of	f -
	(1) 2	(2) $\sqrt{2}$	(3)	$1/\sqrt{2}$	(4	₁₎ $\sqrt{3}$	
11.			us of the pla	anet remains	same and	I mass becomes 4 times,	the
	escape velocity become (1) $4v_e$	es - (2) 2v _e	(3)	Ve	(4	I) Ve	
12.	How many times is esc	ape velocity (V _e),	of orbital ve	elocity (V ₀) for	a satellite	e revolving near earth -	
	(1) $\sqrt{2}$ times	(2) 2 times	(3)	3 times	(4	1) 4 times	
13.	If the radius of a planet		•		-	surface will be -	
		(2) $v_e \propto R^{\sqrt{p}}$		$\frac{\sqrt{\rho}}{2}$		$_{\rm I) \ V_e \ \propto} \frac{1}{\sqrt{ ho R}}$	
14.	(1) $v_e \propto R$ If V, R and g denote reacceleration due to gra	spectively the esc	cape velocity	y from the sur	4) face of the	I) $v_e \propto \sqrt{\rho N}$ e earth radius of the earth,	and
	$(1) \ V = \sqrt{gR}$	(2) V = $\sqrt{\frac{4}{3}}gR^3$	(3)	$V = R^{\sqrt{g}}$	(4	$V = \sqrt{2gR}$	
15.	If the radius of a planet					for both, the escape velo	city
	on the planet will be - (1) 11.2 km/s	(2) 5.6 km/s	(3)	22.4 km/s	(4	I) None	
16.	If the radius and accele (1) 11.2 km/s	ration due to grav (2) 22.4 km/s		e doubled, eso 5.6 km/s		city of earth will become. 4) 44.8 km/s	
17.	If g is the acceleration of from the surface of the					of an object of mass m rai	sed
				1		1	
	(1) mgR	(2) 2mgR	(3)	$\frac{1}{2}$ mgR	(4) ⁴ mgR	
18.	A missile is launched wienergy will be	th a velocity less	than the eso	cape velocity.	The sum o	of kinetic energy and poter	ntial
	(1) positive(3) negative or positive	, uncertain		negative zero			
19.	these are related by :		•			e to the earth's surface, t	hen
	(1) $v_0 = \sqrt{2}v_e$	(2) $V_0 = V_e$	(3)	$v_e = \sqrt{2v_0}$	(4) $v_e = \sqrt{2}v_0$	
20.	An artificial satellite mo Its potential energy is: (1) - E ^o (2) 1.5		orbit around		s a total (l (4) E ^o	kinetic + potential) energy	′ E₀.
	(1) - (2) 1.0	_	(3) 2 -		\ ''/ -		

21.

21.	apart. With what veloce that it may escape out	city should a particle of materials to infinity.	nass m be projected from	vely. Their centres are d distance in the mid point of their centres so
	$(1) \sqrt{\frac{G(M_1 + M_2)}{d}}$	(2) $\sqrt{\frac{2G(M_1 + M_2)}{d}}$	$(3) \sqrt{\frac{4G(M_1 + M_2)}{d}}$	$(4) \sqrt{\frac{GM_1M_2}{d}}$
22.	A satellite has to revol of the satellite in this of (1) 16 km/sec		ircular orbit of radius 8 × (3) 3 km/sec	10 ₃ km. The velocity of projection (4) 7.08 km/sec
23.			e moon is 10. The ratio o rface to that from the mo (3) 4	f g an earth to the moon is 6. The on is approximately (4) 2
24.	and the earth are V_p a then	nd V _e respectively Assum	ning that the radii of the p	th. Escape velocity for the planet lanet and the earth are the same,
	(1) V _P = 10 V _e	(2) $V_P = \sqrt{10} V_e$	$V_P = \frac{V_e}{\sqrt{10}}$	$V_P = \frac{V_e}{10}$
25.	A space shuttle is laur	nched in a circular orbit n		The additional velocity be given to
26.			dy so that it may escape	
	(1) mgR _e	(2) 2mgR _e	$\frac{mgR_e}{5}$	(4) 16
27.		of a body of mass 3kg o	n the surface of a planet	is 54 joule. The escape velocity
	will be (1) 18m/s	(2) 162 m/s	(3) 36 m/s	(4) 6 m/s
28.	The escape velocity f same density will be	rom a planet is v_0 . The ϵ	escape velocity from a p	lanet having twice the radius but
	(1) 0.5 v ₀	(2) v ₀	(3) 2v ₀	(4) 4v ₀
29.	If the kinetic energy of (1) the satellite will es (3) radius of its orbit w	cape into the space.	nd the ear:th is doubled the (2) the satellite will fall (4) radius of its orbit w	down on the earth
30.	The escape velocity from (1) mass of earth	om the earth does not do (2) mass of the body	epend upon (3) radius of earth	(4) acceleration due to gravity
31.	There is no atmosphe (1) it is near the earth (2) it is orbiting around (3) there was no gas a (4) the escape velocity	d the earth at all	s than their root-mean so	_l uare velocity
32.	The escape velocity is	3		
	(1) 2gR	(2) gR	(3) \sqrt{gR}	(4) $\sqrt{2gR}$
33.				by a source S, from A to B, along as I, II and III is W, W, and W

respectively, then



(1)
$$W_1 = W_{11} = W_{111}$$

(2)
$$W_{II} > W_{III} = W_{II}$$

(3)
$$W_{III} = W_{II} > W_{I}$$

(4)
$$W_1 > W_{11} > W_{111}$$

34. The escape velocity of a particle of mass m varies as :

$$(1) m_2$$

$$(3) m_0$$

35. Acceleration due to gravity at earth's surface is g ms $_{-2}$. Find the effective value of gravity at a height of 32 km from sea level : ($R_e = 6400 \text{ km}$)

 $(R_e = 6400 \text{ km})$

(1) 0.5 g ms₋₂

(2) 0.99 g ms₋₂

(3) 1.01 g ms₋₂

(4) 0.90 g ms₋₂

36. Radius of orbit of satellite of earth is R. Its kinetic energy is proportional to :

$$\frac{1}{R}$$

(2)
$$\frac{1}{\sqrt{R}}$$

(3) R

(4)
$$\frac{1}{R^{3/2}}$$

37. A cosmonaut is orbiting earth in a spacecraft at an altitude h = 630 km with a speed of 8 km/s. If the radius of the earth is 6400 km, the acceleration of the cosmonaut is

(1) 9.10 m/s₂

(2) 9.80 m/s₂

(3) 10.0 m/s₂

(4) 9.88 m/s₂

38. A very very large number of particles of same mass m are kept at horizontal distances of 1m, 2m, 4m, 8m and so on from (0,0) point. The total gravitational potential at this point is (addition of G.P. of infinite

terms =
$$\frac{1}{1-r}$$
 where a = first term, r = common ratio):

$$(2) - 3G m$$

$$(4) - 2G m$$

39. A body starts from rest at a point, distance R_0 from the centre of the earth of mass M, radius R. The velocity acquired by the body when it reaches the surface of the earth will be

$$\left(\frac{1}{R} - \frac{1}{R_0}\right)$$

$$\left(2) \ 2 \ \mathsf{GM} \left(\frac{1}{R} - \frac{1}{R_0}\right)$$

$$\sqrt{2} GM \left(\frac{1}{R} - \frac{1}{R_0}\right)$$

$$\sqrt{\left(\frac{1}{R} - \frac{1}{R_0}\right)}$$

40. The relation between the escape velocity from the earth and the velocity of a satellite orbiting near the earth's surface is

$$(1) v_e = 3v$$

(2)
$$v_e = v$$

(3)
$$v_e = 2v$$

$$(4) v_e = v/2$$

SECTION (D): KEPLER'S LAW FOR SATELLITES, ORBITAL SPEED AND ESCAPE SPEED

1. A body attains a height equal to the radius of the earth. The velocity of the body with which it was projected is:

$$\sqrt{\frac{GM}{R}}$$

$$(2)$$
 $\sqrt{\frac{2GM}{R}}$

$$\sqrt{\frac{5}{4}\frac{GM}{R}}$$

$$\sqrt{\frac{3GM}{R}}$$

2. A satellite of mass m is circulating around the earth with constant angular velocity. If radius of the orbit is R₀ and mass of the earth M, the angular momentum about the centre of the earth is

(1)
$$m\sqrt{GMR_0}$$

(2)
$$M\sqrt{GmR_0}$$

$$m\sqrt{\frac{GM}{R_0}}$$

$$M\sqrt{\frac{GM}{R_0}}$$

3. Which of the following quantity is conserved for a satellite revolving around the earth in particular orbit?

- (1) Angular velocity
- (2) Force
- (3) angular momentum (4) Velocity

 $(1) \sqrt{10}$

4.

	(1) $\sqrt{10}$	(2) 100	(3) $10\sqrt{10}$	(4) ¹ /√10
5.	The period of a satellite of radius 4R is -	in a circular orbit of radi	us R is T, the period of a	nother satellite in a circular orbit
	(1) 4T	(2) T/4	(3) 8T	(4) T/8
6.	T is related to the radius	s of the circular orbit as -	•	aken to complete one revolution
	(1) T ∝ r	(2) T ∝ r2	(3) T2 ∝ r3	(4) T ∝ r4
7.		a sphere of mass m from $2\sqrt{\frac{GM}{R}}$		
	()	()	(-)	
8.		om the earth is about 11 e same mean density as (2) 11 km/sec		e velocity from a planet having (4) 15.5 km/sec
9.		2 times that of the earlier		orbit. Its distance from the centre the second orbit is -
	(1) 4.8 hours	(2) $48\sqrt{2}$ hours	(3) 24 hrs	(4) Infinite
10.	speed of the satellite A	is 3V, the speed of the s	atellite B will be	lill 4R and R respectivley. If the
	(1) 12 V	(2) 6 V	(3) 4/3 V	(4) 3/2 V
11.				a planet where acceleration due that of earth will be in km/sec-(4) 53.6
12.	(1) Is the same at all po(2) Is greatest when it is(3) Is greatest when it is			mass of the satellite
13.	planet, whose radius is	n of a satellite around a p 3R but having same der		riod of revolution around another
	$(1) \frac{1}{3\sqrt{3}}$	(2) 3T	(3) 9T	(4) 3√3 T
14.	If Ve and Vo represent orbit of radius R, then -	the escape velocity and	·	ellite corresponding to a circular
		$\sqrt{2}N$ N	$V_e = \frac{1}{\sqrt{2}}V_o$	
15.	(1) Ve = Vo A spherical planet far acceleration due to grav		(3) $\sqrt{2}$ ss M ₀ and diameter D ₀	(4) none of these o. A particle will experience an
	(1) GM ₀ /D ₀₂	(2) 2mGM ₀ /D ₀₂	(3) 4GM ₀ /D ₀₂	(4) GmMo/Do2
16.	the angular velocity of		doubles, a satellite can	from the centre of the planet. If now be in a geostationary orbit

The distance of neptune and saturn from sun are nearly 1013 and 1012 meters respectively. Assuming that $text{hey}$ move in circular orbits, their periodic times will be in the ratio -

 $(3) 10\sqrt{10}$

(4) $1/\sqrt{10}$

	<u>r</u>	r	r	r
	(1) $\frac{1}{2}$	$(2) \frac{r}{2\sqrt{2}}$	(3) $\overline{(4)^{1/3}}$	(4) $\overline{(2)^{1/3}}$
17.	Consider a satellite goi (1) It is a freely falling to (3) It is moving with a control	oody	orbit. Which of the follow (2) It suffers no acceler (4) Its angular moment	
18.	The period of a satellite (1) The mass of the pla (3) The mass of the sa		d a planet is indepent of (2) The radius of the pla (4) All the three parame	anet
19.	A small satellite is revo	olving near earth's surface (2) 11.2 km/sec	e. Its orbital velocity will b (3) 4 km/sec	pe nearly. (4) 6 km/sec
20.	(1) Is the same at all po(2) Is greatest when it i(3) Is greatest when it i			mass of the satellite
21.	If the height of a satellit velocity of the satellite		·	radius of the earth R, the orbital
	(1) gR	(2) gR/2	(3) $\sqrt{g/R}$	(4) \sqrt{gR}
22.	Orbital velocity of an au (1) Mass of the earth (3) Radius of the earth	rtificial satellite does not c	depend upon - (2) Mass of the satellite (4) Acceleration due to	
23.	The time period of a get (1) 24 hours	eostationary satellite is - (2) 12 hours	(3) 365 days	(4) One month
24.	Which one of the following statements regarding artificial satellite of the earth is incorrect - (1) The orbital velocity depends on the mass of the satellite (2) A minimum velocity of 8 km/sec is required by a satellite to orbit quite close to the earth (3) The period of revolution is large if the radius of its orbit is large (4) The height of a geostationary satellite is about 36000 km from earth			
25.	Two identical satellites are at R and 7R away from earth surface, the wrong statement is (R = Radius of earth) (1) Ratio of total energy will be 4 (2) Ratio of kinetic energies will be 4 (3) Ratio of potential energies will be 4 (4) Ratio of total energy will be 4 but ratio of potential and kinetic energies will be 2			
26.	For a satellite escape when escape velocity w	ill be -		an angle of 60° with the vertical,
	(1) 11 km/s	(2) 11 $\sqrt{3}$ km/s	(3) $\frac{11}{\sqrt{3}}$ km/s	(4) 33 km/s
27.	_	stationary satellite from th	ne centre of the earth (Ra	adius R = 6400 km) is nearest
	to - (1) 5 R	(2) 7 R	(3) 10 R	(4) 18 R
28.		lite revolving above Earth o gravity at Earth's surfac		ual to R, radius of Earth, is

	2R	\overline{R}	\overline{R}	\overline{R}
	(1) 2π √ <i>g</i>	$_{(2)}\sqrt{2}\pi\sqrt{\frac{R}{g}}$	(3) 2π [√] <i>g</i>	(4) $8\pi\sqrt{g}$
29.	Given radius of Earth 'I constant. M-Mass of Earth		' the height of a geostation	onary satellite is [G-Gravitational
	(1) +R	(2) – R	(3) – R	(4) None
30.	Distance of geostationalis -	ary satellite from the surf	ace of earth radius (R_e =	6400 km) in terms of $R_{\mbox{\tiny e}}$
	(1) 13.76 R _e	(2) 10.76 R _e	(3) 5.56 R _e	(4) 2.56 R _e
31.	The orbital velocity of a	a planet revolving close to	o earth's surface is -	[-
	(1) $\sqrt{2gR}$	(2) \sqrt{gR}	(3) $\sqrt{\frac{2g}{R}}$	(4) $\sqrt{\frac{g}{R}}$
32.	A satellite moves arourd is M, its total energy is		orbit of radius r with spe	eed v. If the mass of the satellite
		(2) $\frac{1}{2}$ Mv ₂	<u>3</u>	
	(1) $\frac{-2}{2}$ Mv ₂	(2) 2 Mv ₂	(3) 2 Mv ₂	(4) Mv ₂
33.	If satellite is shifted tow (1) Increase	vards the earth. Then tim (2) Decrease	e period of satellite will b (3) Unchanged	ne - (4) Nothing can be said
34.	of satellite A is 3v, ther	go round a planet in circunt speed of satellite B is -	ılar orbits having radii 4R	and R, respectively. If the speed
	$\frac{3v}{2}$	(2) $\frac{4v}{2}$		
	(1) 2	(2) 2	(3) 6v $\frac{1}{3}$ $\frac{1}{3}$	(4) 12v
35.	If gravity field due to a of a satellite near earth	point mass follows g ∝ 's surface and radius of	r^2 insted of r^2 , then t	the relation between time period
	(1) $T_2 \propto r_3$	(2) T ∝ r ₂	(3) T ₂ ∝ r	(4) T ∝ r
36.	A satellite appears to b (1) 35800km (3) 6400km	e at rest when seen from	n the equator. Its height f (2) 358000 km (4) such a satellite can	rom the earth's surface is nearly not exist
37.	(1) it will burn on enteri		•	
38.	(1) the plane of great c	move only in those orbits ircle of earth at any latitude of earth	(2) the plane passing the	with hrough the poles of earth
39.	A satellite launching sta (1) near the equatorial		(2) near the polar region	
40.			4) all locations are equ to be placed at the s	ually good urface of earth for world-wide
	communication betwee (1) 6	en any two locations is - (2) 4	(3) 3	(4) 5

41.

	approximately be:	· ·		,
	(1) 1/2 hr	(2) 1 hr	(3) 2 hr	(4) 4 hr
42.	ejected from the satellite	e such that it just escape	s from the gravitational p	e earth. An object of mass 'm' is ull of the earth. At the time of its
	$\frac{1}{m}V^2$		$\frac{3}{2}mV^2$	
	ejection, the kinetic energy $\frac{1}{2}mV^2$	(2) mV ₂	(3) 2	(4) 2mV ₂
43.	Two satellite of earth, SS ₂ . Which one of the fo (1) The time period of S(2) The potential energi (3) S ₁ and S ₂ are movin (4) The kinetic energies	Illowing statements is trues is four times that of S ₂ es of earth and satellite in g with the same speed of the two satellites are	e : in the two cases are equal	s of S_1 is four times the mass of al
44.	-	atellite revolving nearby		
	(1) $\sqrt{2gR}$	(2) \sqrt{gR}	(3) $\sqrt{g/R}$	(4) $\sqrt{2g/R}$
45.	If the radius of earth is of (1) decrease by 2%	decreased by 1% and ma (2) decrease by 1%	ass remain constant, ther (3) increase by 1%	the acceleration due to gravity (4) increase by 2%
46.	Escape velocity for a ro	cket is 11.2 km/s. If it is	taken to a planet where	the radius and acceleration due
	to gravity is double than (1) 5.6 m/s	n earth, then escape velo (2) 11.2 m/s	ocity will be: (3) 22.4 km/s	(4) 44.2 m/s
47.	Suppose radius of the r	noon's orbit around the	earth is doubled. Its perio	d around the earth will become:
	(1) 1/2 times	(2) $\sqrt{2}$ times	(3) 2 _{2/3} times	(4) 2 _{3/2} times
48.	In case of an orbiting sa (1) its Kinetic Energy de (3) its Mechanical Energy		oit is decreased : (2) its Potential Energy (4) its speed decreases	
49.	A satellite of the earth is revolving in a circular orbit with a uniform speed v. If the gravitational force suddenly disappears, the statellite will (1) Continue to move with velocity v along the original orbit (2) Move with a velocity v, tangentially to the original orbit (3) Fall down with increasing velocity (4) Ultimately come to rest somewhere on the original orbit			
50.	The time period of a sa	tellite of earth is 5 hours	. If the separation betwe	en the earth and the satellite is
		e previous value, the new (2) 80 hour		(4) 20 hour
51.			ally upwards from the sutical, the escape velocity	
	(1) 11 $\sqrt{2}$ km/s	(2) 22 km/s	(3) 11 km/s	(4) 11/ $\sqrt{2}$ m/s
52.	acceleration due to grav	vity on the surface of the	earth, the orbital speed	1/2
		gR	$\frac{gR^2}{R+x}$	$\left(\underline{gR^2}\right)^{1/2}$
	(1) gx	(2) $\frac{\overline{R-x}}{R}$	$\frac{\overline{R}+x}{R}$	(4) $(R+x)$
53.	The time period of an earth (1) the mass of the sate	arın satellite in circular o	rbit is independent of :	(7)

A geostationary satellite orbits around the earth in a circular orbit of radius 36000 km. Then, the time

period of a spy satellite orbiting a few hundred kilometers above the earth's surface (Rearth = 6400 km) will

(1) 2mgR

54.

(2) radius of its orbit

(3) both the mass and radius of the orbit

(4) neither the mass of the satellite nor the radius of its orbit

55.		rth. When both 'd' and 'h		rth is the same as at a depth 'd' the radius of earth, then, which
	$\frac{h}{}$	$\frac{3h}{2}$ (2) d = $\frac{3}{2}$		
	(1) $d = 2$	(2) $d = 2$	(3) d = 2h	(4) d = h
56.	Find the work to be don		al force between them, to	mass 100 kg and radius 10 cm. o take the particle far away from
		(2) $3.33 \times 10_{-10} \text{ J}$		(4) 6.67 × 10 ₋₇ J
57.	and if Millikan's oil drop Electronic charge on the moor	expriment could be perform		earth and the moon respectively es, one will find the ratio
	electronic charge on the earth (1) 1	(2) 0	(3) g _E /g _M	(4) gм/g _E
58.		` ,	, , , ,	earth and its radius is 10 times
00.				escape velocity from the surface
	(1) 11 km s ₋₁	(2) 110 km s ₋₁	(3) 0.11 km s ₋₁	(4) 1.1 km s ₋₁
SECT	ION (E) : THE EART	H AND OTHER PLA	NETS GRAVITY	
1.	they move in circular or	e and saturn from sun are bits, their periodic times		ters respectively. Assuming that
	(1) $\sqrt{10}$	(2) 100	(3) 10	(4) 1/10
2.	The period of a satellite orbit of radius 4R is.	e in a circular orbit of era	adius R is T, the period	of another satellite in a circular
	(1) 4T	(2) T/4	(3) 8T	(4) T/8
3.	Two planets move arour respectively. The ratio		times and the mean rac	dii of the orbits are T_1 , T_2 and r_1 ,
	(1) (r ₁ / r ₂) _{1/2}	(2) r ₁ / r ₂	(3) $(r_1 / r_2)_2$	(4) $(r_1 / r_2)_{3/2}$
4.		n an orbit at a distance o		s 83 minutes. The time period of s surface will be -
	(1) 83 minute	(2) $83 \times \sqrt{8}$ minutes	(3) 664 minutes	(4) 249 minutes
5.		s circulating around the earth M, the angular mon		lar velocity. If radius of the orbit of the earth is -
	,		GM	$ \underline{GM} $
•	(1) m $\sqrt{GMR_0}$	(2) M $\sqrt{GMR_0}$	(3) m $\sqrt{R_0}$	(4) M $\sqrt{R_0}$
6.	A planet revolves aroun sun. The revolution time		nce is 1.588 times the m	ean distance between earth and
	(1) 1.25 years	(2) 1.59 years	(3) 0.89 years	(4) 2 years
•				

If g is the acceleration due to gravity on the earth's surface, the gain in the potential energy of an object

(4) mgR

of mass m raised from the surface of the earth to a height equal to the radius R of the earth, is:

	7.	If mass of a satellite is o	doubled and time period	remain constant the ratio	o of orbit in the two cases will be
 period of revolution will be - (1) 1/2 years (2) √2 years (3) 4 years (4) 8 years 9. A body revolved around the sun 27 times faster then the earth what is the ratio of their radii (1) 1/3 (2) 1/9 (3) 1/27 (4) 1/4 10. The orbital angular momentum of a satellite revolving at a distance r from the centre is L. If the distan is increased to 16r, then the new angular momentum will be - (1) 16 L (2) 64 L (3) 4 (4) 4 L 11. The ratio of the distance of two planets from the sun is 1.38. The ratio of their period of revolution around the sun is - (1) 1.38 (2) 1.38∞ (3) 1.38∞ (4) 1.38∞ (4) 1.38∞ (4) 1.38∞ (1) 1.38∞ (2) 1.38∞ (3) 1.38∞ (4) 1.38∞ (4) 1.38∞ (1) 1.38∞ (2) 1.38∞ (3) 1.38∞ (4) 1.38∞ (4) 1.38∞ (1) 1.38∞ (2) 1.38∞ (3) 1.38∞ (4) 1.		(1) 1 : 2	(2) 1 : 1	(3) 1 : 3	(4) None of these
 A body revolved around the sun 27 times faster then the earth what is the ratio of their radii (1) 1/3 (2) 1/9 (3) 1/27 (4) 1/4 The orbital angular momentum of a satellite revolving at a distance r from the centre is L. If the distan is increased to 16r, then the new angular momentum will be - L (4) 4 L The ratio of the distance of two planets from the sun is 1.38. The ratio of their period of revolution arou the sun is - (1) 1.38 (2) 1.38∞ (3) 1.38∞ (4) 1.38₀ (2) 1.38₀ (2) (3) 1.38∞ (4) 1.38₀ (2) (3) 1.38₀ (2) (4) 1.38₀ Kepler's second law (law of areas) is nothing but a statement of - (1) Work energy theorem (3) Conservation of angular momentum (4) Conservation of linear momentum (3) Conservation of angular momentum (4) Conservation of energy In an elliptical orbit under gravitational force, in general. (1) Tangential velocity is constant (2) Angular velocity is constant (3) Radial velocity is constant (4) Areal velocity is constant What does not change in the field of central force. (1) Potential energy (2) Kinetic energy (3) Linear momentum (4) Angular momentum A planet is moving in an elliptic orbit. If T, V, E, L stand respectivley for its kinetic energy, gravitation potential energy, total energy and magnitude of angular momentum about the centre of force, which the following statements is correct (1) T is conserved (2) V is always positive (3) E is always positive (3) E is always positive (3) E is always positive (4) L is conserved but the direction of vector changes continuously EXERCISES - 2 Three identical stars of mass M are located at the vertices of an equilateral triangle with side L. The spe at which they will move if they all revolve under the influence of one another's gravitational force in circular orbit circumscribing the triangle while still preserving the equilateral triangle:	8.		be -	If the distance between	them becomes double, the new
 (1) 1/3 (2) 1/9 (3) 1/27 (4) 1/4 The orbital angular momentum of a satellite revolving at a distance r from the centre is L. If the distan is increased to 16r, then the new angular momentum will be - L. (4) 4 L. (5) 4 L. (6) 4 L. (7) 4 L. (7) 4 L. (7) 4 L. (8) 4 L. (7) 4 L. (8) 4 L. (8) 4 L. (9) 4 L. (1) 1. The ratio of the distance of two planets from the sun is 1.38. The ratio of their period of revolution arou the sun is - (1) 1.38 (2) 1.38₃₂ (3) 1.38₃₂ (4) 1.38₅ Kepler's second law (law of areas) is nothing but a statement of - (1) Work energy theorem (2) Conservation of linear momentum (3) Conservation of angular momentum (4) Conservation of linear momentum (3) Conservation of angular momentum (4) Conservation of energy In an elliptical orbit under gravitational force, in general. (1) Tangential velocity is constant (2) Angular velocity is constant (3) Radial velocity is constant (4) Areal velocity is constant (4) Areal velocity is constant What does not change in the field of central force. (1) Potential energy (2) Kinetic energy (3) Linear momentum (4) Angular momentum A planet is moving in an elliptic orbit. If T, V, E, L stand respectively for its kinetic energy, gravitation potential energy, total energy and magnitude of angular momentum about the centre of force, which the following statements is correct (1) T is conserved (2) V is always positive (3) E is always negative (4) L is conserved but the direction of vector changes continuously EXERCISE—2 Three identical stars of mass M are located at the vertices of an equilateral triangle with side L. The spe at which they will move if they all revolve under the influence of one another's gravitational force in circular orbit circumscribing the triangle while still preserving the equilateral triangle with side L. The spe at which they will move if they all revolve under the influence of one another's gravitational force in circular orbit circumscribing the triangle while still		(1) 1/2 years	(2) 2 $\sqrt{2}$ years	(3) 4 years	(4) 8 years
is increased to 16fr, then the new angular momentum will be - (1) 16 L (2) 64 L (3) $\frac{L}{4}$ (4) 4 L 11. The ratio of the distance of two planets from the sun is 1.38. The ratio of their period of revolution arou the sun is - (1) 1.38 (2) 1.38 ₃₂ (3) 1.38 _{1/2} (4) 1.38 ₅ 12. Kepler's second law (law of areas) is nothing but a statement of - (1) Work energy theorem (2) Conservation of linear momentum (3) Conservation of angular momentum (4) Conservation of energy 13. In an elliptical orbit under gravitational force, in general. (1) Tangential velocity is constant (2) Angular velocity is constant (3) Radial velocity is constant (4) Areal velocity is constant 14. What does not change in the field of central force. (1) Potential energy (2) Kinetic energy (3) Linear momentum (4) Angular momentum 15. A planet is moving in an elliptic orbit. If T, V, E, L stand respectivley for its kinetic energy, gravitatior potential energy, total energy and magnitude of angular momentum about the centre of force, which the following statements is correct (1) T is conserved (2) V is always positive (3) E is always negative (4) L is conserved but the direction of vector changes continuously EXERCISE—2 1. Three identical stars of mass M are located at the vertices of an equilateral triangle with side L. The spe at which they will move if they all revolve under the influence of one another's gravitational force in circular orbit circumscribing the triangle while still preserving the equilateral triangle with side L. The spe at which they will move if they all revolve under the influence of one another's gravitational force in circular orbit circumscribing the triangle while still preserving the equilateral triangle with side L. The spe at which they will move if they all revolve under the influence of one another's gravitational force in circular orbit circumscribing the triangle while still preserving the equilateral triangle at all 2. Periodic-time of a satellite revolving very near to the surface of the earth	9.				
 The ratio of the distance of two planets from the sun is 1.38. The ratio of their period of revolution arou the sun is - (1) 1.38 (2) 1.38_{xz} (3) 1.38_{xz} (4) 1.38₃ Kepler's second law (law of areas) is nothing but a statement of - (1) Work energy theorem (2) Conservation of linear momentum (3) Conservation of angular momentum (4) Conservation of energy In an elliptical orbit under gravitational force, in general. (1) Tangential velocity is constant (2) Angular velocity is constant (3) Radial velocity is constant (4) Areal velocity is constant What does not change in the field of central force. (1) Potential energy (2) Kinetic energy (3) Linear momentum (4) Angular momentum A planet is moving in an elliptic orbit. If T, V, E, L stand respectivley for its kinetic energy, gravitation potential energy, total energy and magnitude of angular momentum about the centre of force, which the following statements is correct (1) T is conserved (2) V is always positive (3) E is always negative (4) L is conserved but the direction of vector changes continuously Exercise-2 Three identical stars of mass M are located at the vertices of an equilateral triangle with side L. The spe at which they will move if they all revolve under the influence of one another's gravitational force in circular orbit circumscribing the triangle while still preserving the equilateral triangle: (1) √D/L (2) √D/L (3) √D/L (4) not possible at all Periodic-time of a satellite revolving very near to the surface of the earth is - (ρ is density of earth) (1) Proportional to D/D (2) Proportional to T/D (3) Proportional p (4) does not depend on ρ. A satellite is moving round the earth. In order to make it move to infinity, its velocity must be increas by (1) 20% 	10.			ntum will be -	m the centre is L. If the distance
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 Three identical stars of mass M are located at the vertices of an equilateral triangle with side L. The spe at which they will move if they all revolve under the influence of one another's gravitational force in circular orbit circumscribing the triangle while still preserving the equilateral triangle: \[\sum_{\sum_{i}} \frac{2 GM}{L} \] (1) \[\sum_{i} \frac{Q}{L} \] (2) \[\sum_{i} \frac{Q}{L} \] (3) \[\sum_{i} \frac{GM}{L} \] (4) not possible at all Periodic-time of a satellite revolving very near to the surface of the earth is - (\rho is density of earth) \[\frac{1}{\sum_{\rho}} \] (1) Proportional to \[\frac{1}{\rho} \] (2) Proportional to \[\frac{1}{\sum_{\rho}} \] (3) Proportional \(\rho \) (4) does not depend on \(\rho\). A satellite is moving round the earth. In order to make it move to infinity, its velocity must be increas by (1) 20% (2) it is impossible to do so 		Exercise	-2		
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Periodic-time of a satellite revolving very near to the surface of the earth is - (ρ is density of earth) $\frac{1}{\rho}$ (1) Proportional to $\frac{1}{\rho}$ (2) Proportional to $\frac{1}{\sqrt{\rho}}$ (3) Proportional ρ (4) does not depend on ρ . A satellite is moving round the earth. In order to make it move to infinity, its velocity must be increas by (1) 20% (2) it is impossible to do so	1.	at which they will move circular orbit circumscri	e if they all revolve unde bing the triangle while st	er the influence of one a ill preserving the equilate	another s gravitational force in a eral triangle :
(1) Proportional to $\frac{1}{\rho}$ (2) Proportional to $\frac{1}{\sqrt{\rho}}$ (3) Proportional ρ (4) does not depend on ρ . A satellite is moving round the earth. In order to make it move to infinity, its velocity must be increas by (1) 20% (2) it is impossible to do so					
 A satellite is moving round the earth. In order to make it move to infinity, its velocity must be increas by (1) 20% (2) it is impossible to do so 	2.	1	1		
by (1) 20% (2) it is impossible to do so		(1) Proportional to ρ	(2) Proportional to $\sqrt{\rho}$	(3) Proportional ρ	(4) does not depend on ρ .
(1) 20% (2) it is impossible to do so	3.	_	und the earth. In order to	o make it move to infinity	y, its velocity must be increased
				(2) it is impossible to de	0 S0

	(1) remain sam	e (2) increase by	4% (3) decrea	ase by 4%	(4) increase by 2%	
5.			the satellite as a re (2) gravita		ange al energy	Which one
6.	because its (1) period is gre (2) period is les (3) period has r	ite transmits the television eater than the period of rota is than the period of rota ino relation with the perion and to the period of rotati	otation of the earth tion of the earth ab d of the earth abou	out its axis t its axis	one part of the world	to another
7.	If the universal still maintain its (1) radius		_	•	ne, then a satellite in o	
	(1) Tadius	(2) tangential speed	(3) ariguia	ai inomentum	(4) period of revoluti	OH
8.	compared to th (1) The acceler (2) The angula remains consta (3) The total me	moving in an elliptical e mass of the earth: ation of S is always direct momentum of S about ant echanical energy of S vanomentum of S remains	cted towards the ce the centre of the cries periodically wit	entre of the ear earth changes h time	rth	-
9.		lves round the earth 13 t, then the ratio of masse (2) 3.56 × 106		will be : 10 ₇	un-earth distance to e (4) 3.56 × 10₃	arth-moon
10.	The earth is rev A will be	olving round the sun in a	n elliptical orbit. If	$\frac{\partial H}{\partial B} = x$, the	ratio of speeds of ear	th at B and
		A	0	В		
	(1) x	(2) \sqrt{x}	(3) x ₂		$(4) x\sqrt{x}$	
11.	increased to 4 to	d of a satellite of earth times the previous value (2) 80 h				satellite is
12.	(1) 10 h If two spheres	of same masses and ra		contact, then		n between
	them will be pro	oportional to (for a given	density ρ) :			
	(1) r ²	(2) r ³	(3) r ⁶		(4) r ⁴	
13.	Assume the ea	orth to be a sphere of rac	dius R. If g is the a	cceleration du	ie to gravity at any po	oint on the

If the radius of earth is to decrease by 4% and its density remains same, then its escape velocity will

earth's surface, the mass of the earth is:

$$\frac{gR}{G}$$

$$\frac{g^2R}{G}$$

$$\frac{gR^2}{G}$$

$$(4) \frac{g^2 R}{G}$$

14. Energy required to transfer a 400 kg satellite in a circular orbit of radius 2 R to a circular orbit of radius 4 R, where R is the radius of the earth. [Given $q = 9.8 \text{ ms}_{-2}$, $R = 6.4 \times 10_6 \text{ m}$]

$$(1) 1.65 \times 10_9 J$$

(2)
$$3.13 \times 10_9 \text{ J}$$

$$(3) 6.26 \times 10_9 H$$

$$(4) 4.80 \times 10_9 J$$

15. Suppose the gravitational force varies inversely as the 4th power of the distance. If a satel lite describes a circular orbit of radius R under the influence of this force, then the time period T of the orbit is proportional to

16. A double star system consists of two stars A and B which have time period T_A and T_B. Radius R_A and R_B and mass M_A and M_B. Choose the correct option.

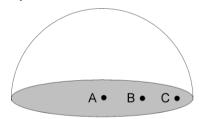
(1) If
$$T_A > T_B$$
 then $R_A > R_B$

(2) If
$$T_A > T_B$$
 then $M_A > M_B$

$$\left(\frac{T_A}{T_B}\right)^2 = \left(\frac{R_A}{R_B}\right)^3$$
 (3)

(4)
$$T_A = T_B$$

17. Mass M is uniformly distributed only on curved surface of a thin hemispherical shell. A, B and C are three points on the circular base of hemisphere, such that A is the centre. Let the gravitational potential at points A, B and C be VA, VB, Vc respectively. Then



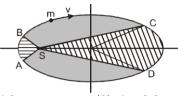
- (1) V_A> V_B>V_C
- (2) Vc> Vb>VA
- (3) $V_B > V_A$ and $V_B > V_C$
- (4) $V_A = V_B = V_C$

Exercise-3

PART - I: NEET / AIPMT QUESTION (PREVIOUS YEARS)

1. The figure shows elliptical orbit of a planet m about the sun S. The shaded area SCD is twice the shaded area SAB. If t1 is the time for the planet to move from C to D and t2 is the time to move from A to B, then:

[AIPMT 2009]



(1) t1 > t2

$$(2) t1 = 4t2$$

(3) t1 = 2t2

$$(4) t1 = t2$$

2. A particle of mass M is situated at the center of a spherical shell of same mass and radius a. The

gravitational potential at a point situated at $\frac{a}{2}$ distance from the centre, will be **[AIPMT 2010]** 3GM 2GM GM 4GM

 $-\frac{3GM}{a}$

 $-\frac{2GM}{a}$

 $-\frac{GM}{a}$

$$-\frac{4GM}{a}$$

3. The additional kinetic energy to be provided to a satellite of mass m revolving around a planet of mass M, to transfer it from a circular orbit of radius R1 to another of radius R2 (R2 > R1) is [AIPMT 2010]

 $GmM\left(\frac{1}{R_{1}^{2}}-\frac{1}{R_{2}^{2}}\right) \qquad GmM\left(\frac{1}{R_{1}}-\frac{1}{R_{2}}\right) \qquad (3) \qquad 2GmM\left(\frac{1}{R_{1}}-\frac{1}{R_{2}}\right) \qquad (4) \qquad \frac{1}{2}GmM\left(\frac{1}{R_{1}}-\frac{1}{R_{2}}\right)$

4. The radii of circular orbits of two satellites A and B of the earth, are 4R and R, respectively. If the speed of satellite A is 3V, then the speed of satellite B will be [AIPMT 2010, Screening]

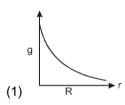
 $\frac{3V}{4}$

(2) 6V

(3) 12 V

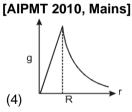
 $(4) \frac{3V}{2}$

5. The dependence of acceleration due to gravity g on the distance r from the centre of the earth, assumed to be a sphere of radius R of uniform density is as shown in figures below. The correct figure is.



(2) R

g R



6. A planet moving along an elliptical orbit is closest to the sun at a distance r1 and farthest away at a

 $\underline{v_{\mathsf{l}}}$

distance of r2. If v1 and v2 are the linear velocities at these points respectively, then the ratio v_2 is :

[AIPMT (Scr) 2010]

(1) (r1/r2)2

(2) r2/r1

(3) (r2/r1)2

(4) r1/r2

- 7. A body projected vertically from the earth reaches a height equal to earth's radius before returning to the earth. The power exerted by the gravitational force is greatest: [AIPMT (Scr) 2010]
 - (1) at the highest position of the body
 - (2) at the instant just before the body hits the earth
 - (3) it remains constant all through
 - (4) at the instant just after the body is projected
- 8. A particle of mass m is thrown upwards from the surface of the earth, with a velocity u. The mass and the radius of the earth are , respectively, M and R. G is gravitational constant and g is acceleration due to gravity on the surface of the earth. The minimum value of u so that the particle does not return back to earth, is:

 [AIPMT (Mains) 2011]

 $\sqrt{\frac{2GM}{R}}$

 $\sqrt{\frac{2GM}{R^2}}$

(3) $\sqrt{2gR^2}$

 $\sqrt{\frac{2GM}{R^2}}$

9. A particle of mass M is situated at the centre of spherical shell of mass and radius a. The magnitude of the gravitational potential at a point situated at a/2 distance from the centre will be:

[AIPMT(Mains)2011]

 $\frac{2GM}{a}$

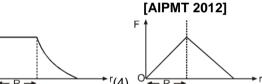
 $\frac{3GM}{a}$

 $\frac{4GN}{a}$

 $\frac{GM}{a}$

10. Which one of the following plots represents the variation of gravitational field on a particle with distance r due to a thin spherical shell of radius R? (r is measured from the centre of the spherical shell)





11. The height at which the weight of a body becomes 1/16th, its weight on the surface of earth (radius R), is:

[AIPMT 2012]

(1) 5R

(2) 15R

(3) 3R

(4) 4R

A spherical planet has a mass MP and diameter DP. A particle of mass m falling freely near the surface of this planet will experience an acceleration due to gravity, equal to:

[AIPMT 2012]

(1) 4GMP/DP2

(2) GMPm/DP2

(3) GMP/DP2

(4) 4GMPm/DP2

A geostationary satellite is orbiting the earth at a height of 5R above that surface of the earth, R being the radius of the earth. The time period of another satellite in hours at a height of 2R from the surface of the earth is:

[AIPMT 2012]

(1) 5

(2) 10

(3) $6\sqrt{2}$

 $\frac{6}{\sqrt{2}}$

A body of mass 'm' is taken from the earth's surface to the height equal to twice the radius (R) of the earth. The change in potential energy of body will be: [NEET-2013; 3/180, -1]

 $\frac{2}{3}mgR$

(2) 3mgR

 $\frac{1}{3}mgR$

(4) mg2R

15. Infinite number of bodies, each of mass 2 kg are situated on x-axis at distances 1m, 2m, 4m, 8m, respectively, from the origin. The resulting gravitational potential due to this system at the origin will be:

[NEET-2013; 3/180, -1]

 $-\frac{8}{3}G$

 $-\frac{4}{3}G$

(3) –4G

(4) - G

16. A block hole is an object whose gravitational field is so strong that even light cannot escape from it. To what approximate radius would earth (mass = 5.98×1024 kg) have to be compressed to be a black hole?

[AIPMT-2014]

(1) 10-9 m

(2) 10-6 m

(3) 10-2 m

(4) 100 m

17. Dependence of intensity of gravitational field (E) of earth with distance (r) from centre of earth is correctly represented by:

[AIPMT-2014]

(1) R

(2) R F

(3) R



18. Kepler's third law states that square of period of revolution (T) of a planet around the sun, is proportional to third power of average distance r between sun and planet i.e. T2 = Kr3 here K is constant. If the masses of sun and planet are M and m respectively then as per Newton's law of gravitation force of attraction between them is [AIPMT-2015]

GMm

 $F = r^2$, here G is gravitational constant. The relation between G and K is described as:

(1) GMK = $4\pi 2$

(2) K = G

(3) K = $\frac{1}{G}$

(4) $GK = 4\pi 2$

19. A remote - sensing satellite of earth revolves in a circular orbit at a height of 0.25×106 m above the surface of earth. If earth's radius is 6.38×106 m and g = 9.8 ms-2, then the orbital speed of the satellite is:

[AIPMT-2015]

(1) 8.56 km s-1

(2) 9.13 km s-1

(3) 6.67 km s-1

(4) 7.76 km s-1

20. At what height from the surface of earth the gravitational potential and the value of g are -5.4×10^7 J kg⁻² and 6.0 ms⁻² respectively? Take the radius of earth as 6400 km. **[AIPMT_2016]**

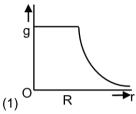
(1) 2000 km

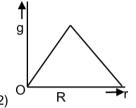
(2) 2600 km

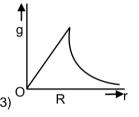
(3) 1600 km

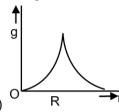
(4) 1400 km

- 21. A satellite S is moving in an elliptical orbit around the earth. The mass of the satellite is very small compared to the mass of the earth. Then, [AIPMT 2016]
 - (1) the total mechanical energy of S varies periodically with time.
 - (2) the linear momentum of S remains constant in magnitude.
 - (3) the acceleration of S is always directed towards the centre of the earth.
 - (4) the angular momentum of S about the centre of the earth changes in direction, but its magnitude remains constant.
- 22. The ratio of escape velocity at earth (ve) to the escape velocity at a planet (vp) whose radius and mean density are twice as that of earth is: [AIPMT-2016]
 - (1) 1 : $\sqrt{2}$
- (2)1:2
- (3) 1 : $2\sqrt{2}$
- (4)1:4
- 23. Starting from the centre of the earth having radius r, the variation of g (acceleration due to gravity) is shown by [NEET 2016-17]









- 24. A satellite of mass m is orbiting the earth (of radius R) at a height h from its surface. The total energy of the satellite in terms of g₀, the value of acceleration due to gravity at the earth's surface, is [NEET 2017]
 - $(1) \frac{30}{R+h}$ (2) $\frac{11.30}{2(R+h)}$
- $\frac{mg_0R^2}{2(R+h)}$
- A physical quantity of the dimensions of length that can be formed out of c, G and $^{4\pi}$ $^{\epsilon}_{0}$ is [c is velocity 25. of light, G is universal constant of gravitation and e is charge]: [NEET 2017]

$$\frac{1}{c^{2}} \left[G \frac{e^{2}}{4\pi \in_{0}} \right]^{1/2} \qquad (2) c^{2} \left[G \frac{e^{2}}{4\pi \in_{0}} \right]^{1/2} \qquad (3) \frac{1}{c^{2}} \left[\frac{e^{2}}{G4\pi \in_{0}} \right]^{1/2} \qquad (4) \frac{1}{c} G \frac{e^{2}}{4\pi \in_{0}}$$

$$(2) c^{2} \left[G \frac{e^{2}}{4\pi \in_{0}} \right]^{1/2}$$

$$\frac{1}{c^2} \left[\frac{e^2}{G4\pi \in_0} \right]^{1/2}$$

$$\frac{1}{c}G\frac{e^2}{4\pi \in_0}$$

- 26. Suppose the charge of a proton and an electron differ slightly. One of them is -e, the other is $(e + \Delta e)$. If the net of electrostatic force and gravitational force between two hydrogen atoms placed at a distance d (much greater than atomic size) apart is zero, then Δe is of the order of [Given mass of hydrogen $m_h = 1.67 \times 10^{-27} \text{ kg}$ [NEET-2017]
 - (1) 10⁻²⁰ C
- (2) 10⁻²³ C
- (3) 10⁻³⁷ C
- (4) 10⁻⁴⁷ C
- 27. Two astronauts are floating in gravitational free space after having lost contact with their spaceship. The [NEET-2017]
 - (1) keep floating at the same distance between them
 - (2) move towards each other
 - (3) move away from each other
 - (4) will become stationary

	in the figure. Then	[NEET 2018]
	Kc, respectively. AC is the major axis and SB is perpendicular to AB at the po-	sition of the Sun S as shown
28.	The kinetic energies of a planet in an elliptical orbit about the Sun, at position	s A, B and C are K_A , K_B and



- (1) $K_A < K_B < K_C$
- (2) $K_B > K_A > K_C$
- (3) $K_B < K_A < K_C$
- (4) $K_A > K_B > K_C$
- 29. If the mass of the Sun were ten times smaller and the universal gravitational constant were ten times larger in magnitude, which of the following is not correct? [NEET 2018]
 - (1) Raindrops will fall faster.
 - (2) 'g' on the Earth will not change
 - (3) Time period of a simple pendulum on the Earth would decrease.
 - (4) Walking on the ground would become more difficult.
- 30. The work done to raise a mass m from the surface of the earth to a height h, which is equal to the radius of the earth, is:

 [NEET-2019-I]
 - $\frac{3}{2}$ mgR (2) mgR (3) 2 mgR (4) $\frac{1}{2}$ mgR
- 31. The time period of a geostationary satellite is 24 h, at a height $6R_E$ (R_E is radius of earth) from surface of earth. The time period of another satellite whose height is 2.5 R_E from surface will be :

[NEET_2019-II]

- (1) $6\sqrt{2}h$
- (2) $12\sqrt{2} h$
- (3) $\frac{24}{2.5}$ h
- $\frac{12}{2.5}h$
- 32. Assuming that the gravitational potential energy of an object at infinity is zero, the change in potential energy (final initial) of an object of mass m, when taken to a height h from the surface of earth (of radius R), is given by:

 [NEET_2019 II]

PART - II : JEE (MAIN) / AIEEE PROBLEMS (PREVIOUS YEARS)

3

- 1. The height at which the acceleration due to gravity becomes $\frac{9}{9}$ (where g = the acceleration due to gravity on the surface of the earth) in terms of R, the radius of the earth, is [AIEEE-2009, 4/144]
 - $\frac{R}{\sqrt{2}}$
- $\frac{R}{2}$
- (3) $\sqrt{2} R$
- (4) 2R
- 2. Two bodies of masses m and 4 m are placed at a distance r. The gravitational potential at a point on the line joining them where the gravitational field is zero is : [AIEEE 2011, 4/120, -1]
 - $-\frac{4Gm}{r} \qquad \qquad -\frac{6Gm}{r} \qquad \qquad -\frac{9Gm}{r}$ (1) zero (2) $\frac{-\frac{9Gm}{r}}{r}$

3.	Two particles of equal mass 'm' go around a circle of radius R under the action of their mutual gravitational
	attraction. The speed of each particle with respect to their centre of mass is:

[AIEEE 2011, 11 May; 4, -1]

$$(1) \sqrt{\frac{Gm}{4R}}$$

(2)
$$\sqrt{\frac{Gm}{3R}}$$

$$(3)$$
 $\sqrt{\frac{Gm}{2R}}$

$$\sqrt{\frac{Gm}{R}}$$

What is the minimum energy required to launch a satellite of mass m from the surface of a planet of mass 4. M and radius R in a circular orbit at an altitude of 2R? [JEE-Main 2013; 4/120, -1]

5GmM 6R(1)

$$(2) \frac{2GmM}{3R}$$

$$\frac{GmM}{2R}$$

$$\frac{GmM}{3R}$$

5. Four particles, each of mass M and equidistant from each other, move along a circle of radius R under the action of their mutual gravitational attration, the speed of each particle is:[JEE-Main 2014;4/120,-1]

$$\sqrt{2\sqrt{2}\frac{GM}{R}}$$

(2)
$$\sqrt{2\sqrt{2}\frac{GM}{R}}$$
 (3) $\sqrt{\frac{GM}{R}(1+2\sqrt{2})}$ (4) $\frac{1}{2}\sqrt{\frac{GM}{R}(1+2\sqrt{2})}$

$$\frac{1}{2}\sqrt{\frac{GM}{R}}\left(1+2\sqrt{2}\right)$$

6. From a solid sphere of mass M and radius R, a spherical portion of radius R/2 is removed, as shown in the figure. Taking gravitational potential V = 0 at $r = \infty$, the potential at the centre of the cavity thus formed is:(G = gravitational constant) [JEE(Main)-2015; 4/120, -1]



$$\frac{-GM}{2R}$$

$$(2) \frac{-GM}{R}$$

$$(3) \frac{-2GM}{3R}$$

$$\frac{-2GM}{R}$$

7. If the angular momentum of a planet of mass m, moving around the sun in a circular orbit is L, about the center of the Sun, its areal velocity is: [JEE(Main)-2019]

(1) m

	4L
(2)	m

8. The energy required to take a satellite to a height 'h' above Earth surface (radius or Earth = 6.4×10^3 km) is E₁ and kinetic energy required for the satellite to be in a circular orbit at this height is E₂. The value of h for which E₁ and E₂ are equal is: [JEE(Main)-2019]

 $(1) 1.28 \times 10^4 \text{ km}$

$$(2) 6.4 \times 10^3 \text{ km}$$

(3)
$$3.2 \times 10^3$$
 km

$$(4) 1.6 \times 10^3 \text{ km}$$

9. A satellite i1s moving with a constant speed v in circular orbit around the earth. An object of mass 'm' is ejected from the satellite such that it just escapes from the gravitational pull of the earth. At the time of ejection, the kinetic energy of the object is: [JEE(Main)-2019]

(1) $\frac{3}{2}$ mv²

$$\frac{1}{2}$$
mv²

 $(1) 3.8 \times 10^4 \text{ m/s}$

10.

11.	the radius of the earth	n. Assuming that the effec	t of earth's atmosphere o	ace, such that h< <r [jee(main)-2019]<="" be="" can="" e="" earth="" field="" gravitational="" is="" is:="" minimum="" neglected.="" of="" r="" th="" the="" where=""></r>				
	$(1) \sqrt{gR} \left(\sqrt{2} - 1 \right)$	(2) $\sqrt{\frac{gR}{2}}$	(3) $\sqrt{2gR}$	(4) √gR				
12.		·	·	lues for the Earth. The period of lation of the same pendulum on [JEE(Main)-2019]				
	$(1)^{\frac{3}{2}}$ s	(2) $\frac{2}{\sqrt{3}}$ s	$\frac{\sqrt{3}}{2}s$	$(4) 2\sqrt{3} s$				
13.		n L extends from x = a to x s per unit length of the roo	J	al force it exerts on a point mass [JEE(Main)-2019]				
	(1) $Gm\left[A\left(\frac{1}{a+L}-\frac{1}{a}\right)\right]$		$\operatorname{Gm}\left[A\left(\frac{1}{a+L}-\frac{1}{a}\right)+BL\right]$					
	$\operatorname{Gm}\left[A\left(\frac{1}{a} - \frac{1}{a+L}\right)\right]$	+BL	$\operatorname{Gm}\left[A\left(\frac{1}{a}-\frac{1}{a+L}\right)-\right]$	BL				
14.	same mass, falling to	wards the earth, collides meteorite are the same,	with the satellite complet	of the earth. A meteorite of the tely inelastically. The speeds of the subsequent motion of the [JEE(Main)-2019]				
	(1) In the same circula(3) in an elliptical orbit		(2) such that it escapes(4) in a circular orbit of	•				
15.	Two satellites, A and I	B, have masses m and 2m	n respectively. A is in a ci	ircular orbit of radius R, and B is T_A				
	in a circular orbit of ra	dius 2R around the earth.	The ratio of their kinetic	-				
	(1) 1	(2) $\sqrt{\frac{1}{2}}$	(3) 2	$(4) \frac{1}{2}$				

Two stars of masses 3×10^{31} kg each, and at distance 2×10^{11} m rotate in a plane about their common centre of mass O. A meteorite passes through O moving perpendicular to the star's rotation plane. In order to escape from the gravitational field of this double star, the minimum speed that meteorite should

 $(3) 2.8 \times 10^5 \text{ m/s}$

[JEE(Main)-2019]

 $(4) 2.4 \times 10^4 \text{ m/s}$

have at O is (Take Gravitational constant $G = 6.67 \times 10^{-11} \text{ Nm}^2 \text{ kg}^{-2}$) What is the order of energy of the gas due to its thermal motion?

 $(2) 1.4 \times 10^5 \text{ m/s}$

Answers

						EVED							
						EXER	CISE	- 1					
SECT	(2)): 2.	(1)	2	(4)	4.	(2)	5.	(2)	6.	(2)	7.	(1)
1. 8.	(3)	2. 9.	(1) (3)	3. 10.	(4) (1)	4.	(3)	5.	(3)	0.	(3)	7.	(1)
	TION (B)		(3)	10.	(1)								
1.	(2)	, . 2.	(2)	3.	(1)	4.	(3)	5.	(1)	6.	(2)	7.	(1)
8.	(2)	9.	(4)	10.	(3)	11.	(1)	12.	(3)	13.	(4)	14.	(2)
15.	(2)	16.	(2)	17.	(4)	18.	(3)	19.	(1)	20.	(1)	21.	(4)
22.	(2)	23.	(3)	24.	(3)	25.	(4)	26.	(4)	27 .	(1)	28.	(2)
29.	(1)	30.	(1)	31.	(2)								
	TION (C)												
1.	(2)	2.	(4)	3.	(2)	4.	(3)	5.	(3)	6.	(1)	7.	(3)
8.	(1)	9.	(2)	10.	(2)	11.	(2)	12.	(1)	13.	(2)	14.	(4)
15.	(3)	16.	(2)	17.	(3)	18.	(2)	19.	(4)	20.	(3)	21.	(3)
22.	(4)	23.	(2)	24.	(2)	25.	(3)	26.	(3)	27.	(4)	28. 25	(3)
29. 36.	(1) (1)	30. 37.	(2)	31. 38.	(4)	32. 39.	(4)	33. 40.	(1)	34.	(3)	35.	(2)
	() [ION (D)		(1)	30.	(4)	39.	(3)	40.	(2)				
1.	(1)	, . 2.	(1)	3.	(3)	4.	(3)	5.	(3)	6.	(3)	7.	(1)
8.	(1)	9.	(2)	10.	(2)	11.	(3)	12.	(2)	13.	(1)	14.	(2)
15.	(3)	16.	(3)	17.	(2)	18.	(3)	19.	(1)	20.	(2)	21.	(4)
22.	(2)	23.	(1)	24.	(1)	25.	(4)	26.	(1)	27.	(2)	28.	(2)
29.	(3)	30.	(3)	31.	(2)	32.	(1)	33.	(2)	34.	(3)	35.	(2)
36.	(1)	37.	(2)	38.	(1)	39.	(1)	40.	(3)	41.	(3)	42.	(2)
43.	(3)	44.	(2)	45.	(4)	46.	(3)	47.	(4)	48.	(3)	49.	(2)
50.	(3)	51.	(3)	52.	(4)	53.	(1)	54.	(2)	55.	(3)	56.	(4)
57.	(1)	58.	(2)										
	TION (E)		(2)	2	(4)	4	(2)	_	(4)	c	(4)	7	(2)
1.	(3)	2. 9.	(3)	3. 10.	(4)	4. 11.	(3) (2)	5. 12.	(1)	6. 13.	(4)	7. 14.	(2)
8. 15.	(2) (3)	9.	(2)	10.	(4)	11.	(2)	12.	(3)	13.	(4)	14.	(4)
13.	(3)					EXER	CISE	- 2					
1.	(2)	2.	(2)	3.	(4)	4.	(3)	<u>-</u> 5.	(2)	6.	(4)	7.	(3)
8.	(1)	9.	(1)	10.	(1)	11.	(3)	12.	(4)	13.	(3)	14.	(2)
15.	(2)	16.	(4)	17.	(4)		(0)		(· /		(0)		(-/
	. ,		. ,		. ,	EXER	CISE	- 3					
							RT – I						
1.	(3)	2.	(1)	3.	(4)	4.	(2)	5.	(4)	6.	(2)	7.	(2)
8.	(1)	9.	(2)	10.	(2)	11.	(3)	12.	(1)	13.	(3)	14.	(1)
15.	(3)	16.	(3)	17.	(1)	18.	(1)	19.	(4)	20.	(2)	21.	(3)
22.	(3)	23.	(3)	24.	(3)	25.	(1)	26.	(3)	27.	(2)	28.	(4)
29.	(2)	30.	(4)	31.	(1)	32.	(2)						
_		_					RT – II	_		_		_	
1.	(4)	2.	(4)	3.	(1)	4.	(1)	5.	(4)	6.	(2)	7.	(3)
8.	(3)	9.	(4)	10.	(3)	11.	(1)	12.	(4)	13.	(3)	14.	(3)
15.	(1)												